

# Principles Of Helicopter Aerodynamics

Principles Of Helicopter Aerodynamics Principles of Helicopter Aerodynamics Understanding the principles of helicopter aerodynamics is essential for grasping how these remarkable machines achieve flight. Helicopters operate through complex aerodynamic interactions that allow vertical takeoff and landing, hovering, and advanced maneuvering capabilities. This article explores the fundamental principles governing helicopter aerodynamics, providing insights into the physics behind rotary-wing flight, key components involved, and the operational nuances that make helicopters versatile aircraft.

Fundamentals of Helicopter Aerodynamics Helicopter aerodynamics revolves around the interaction between the rotor blades and the surrounding air. Unlike fixed-wing aircraft that generate lift primarily through forward motion, helicopters rely on rotating blades to produce the necessary lift and thrust. The core principles include lift generation, rotor blade dynamics, induced flow, and blade element theory.

Lift Generation in Helicopters Lift is the force that opposes gravity and enables the helicopter to ascend. In rotorcraft, lift is generated by the rotor blades acting as rotating wings. The key factors influencing lift include:

- Blade Angle of Attack (AoA): The angle between the blade chord line and the relative airflow. Increasing AoA increases lift until stall conditions are reached.
- Rotor RPM: Higher rotational speeds increase the relative airflow over blades, enhancing lift.
- Air Density: Denser air provides more lift; thus, altitude and weather conditions impact lift capacity.
- Blade Shape and Airfoil: Aerodynamically optimized blades produce greater lift efficiently. The Bernoulli's principle and Newton's third law underpin lift production, with airfoil shape and angle of attack manipulating pressure differences and reactive forces to generate lift.

Blade Element Theory and Momentum Theory Understanding how blades produce lift involves two primary theories:

1. Blade Element Theory: Divides each rotor blade into small sections (elements). Each element's lift and drag are calculated based on local conditions, then integrated along the blade span to determine overall performance.
2. Momentum Theory: Considers the rotor as an actuator disc that imparts a downward acceleration to the air, creating a pressure difference that results in lift. It relates the induced velocity of airflow through the rotor to the thrust produced.

Combining these theories provides a comprehensive picture of rotor aerodynamics and helps in optimizing blade design.

Key Components Influencing Helicopter Aerodynamics Several critical parts of a helicopter influence its aerodynamic behavior:

- Rotor Blades Rotor blades are the primary lifting surfaces. Their design determines the efficiency and maneuverability of the helicopter. Important features include:
  - Airfoil Shape: Aerodynamically optimized to maximize lift and minimize drag.
  - Blade Twist: The blade's angle varies along its length to maintain a consistent angle of attack and lift distribution.
  - Blade Pitch Control: Adjusts the angle of attack to control lift and torque.
- Swashplate Assembly The swashplate allows for cyclic and collective pitch adjustments, changing the blade pitch angle during rotation:
  - Collective Pitch: Alters pitch angle uniformly to

increase or decrease lift. - **Cyclic Pitch:** Varies pitch cyclically during rotation to control the helicopter's tilt and directional movement. **Fuselage and Tail Rotor - Fuselage Aerodynamics:** The body shape affects drag and stability. - **Tail Rotor:** Provides anti-torque force to counteract the main rotor's reactive torque, ensuring stable yaw control. **Principles of Helicopter Flight Dynamics** Understanding how helicopters achieve various flight modes involves examining the interactions between aerodynamics, control inputs, and rotor behavior. **Hovering** In hover, the rotor produces just enough lift to counteract gravity. Key factors include: - **Equal Lift Distribution:** The rotor must generate uniform lift across the rotor disc. - **Induced Flow:** Downward airflow caused by the rotor affects lift and power requirements. - **Power Balance:** Power supplied to the rotor matches the power lost to aerodynamic drag and induced flow. Maintaining a stable hover requires precise control of blade pitch and rotor RPM. **Forward Flight** When moving forward, the rotor disc becomes asymmetric: - **Relative Wind:** The 3 advancing blade experiences higher relative wind speed, producing more lift. - **Autorotation of the Retreating Blade:** The retreating blade experiences lower relative wind and can produce less lift, risking stall if not compensated. - **Tilted Rotor Plane:** The rotor disk tilts forward, generating a horizontal component of lift that propels the helicopter. Achieving efficient forward flight involves adjusting blade pitch (collective and cyclic) and rotor speed. **Vertical Ascent and Descent** - **Ascent:** Increasing collective pitch enhances lift, requiring more power. - **Descent:** Decreasing collective reduces lift; controlled descent involves managing rotor speed and blade pitch to prevent excessive speed or loss of control. **Advanced Aerodynamic Phenomena in Helicopters** Several complex aerodynamic phenomena influence helicopter performance and safety. **Blade Stall and Vortexing** - **Blade Stall:** Occurs when the angle of attack exceeds the critical limit, causing airflow separation and loss of lift. - **Vortexing:** The formation of vortices at blade tips (tip vortices) increases drag and reduces efficiency. Design features like blade twist and swept tips help mitigate these issues. **Retreating Blade Stall and Dissymmetry of Lift** - **Retreating Blade Stall:** During forward flight, the retreating blade experiences lower relative wind speed, risking stall. - **Dissymmetry of Lift:** The difference in lift between advancing and retreating blades. Countered by blade flapping and cyclic pitch adjustments. **Ground Effect** When operating close to the ground, airflow patterns change, reducing induced drag and increasing lift efficiency, which is particularly relevant during takeoff and landing. **Conclusion** The principles of helicopter aerodynamics encompass a broad and intricate set of physics that enable rotary-wing aircraft to perform complex maneuvers. From fundamental lift generation and blade dynamics to advanced phenomena like vortex formation and dissymmetry of lift, a thorough understanding of these principles is vital for helicopter design, operation, and safety. Continuous innovations in aerodynamics and blade technology have enhanced helicopter performance, making them versatile and invaluable tools in transportation, rescue, military, and industrial applications. By mastering the principles outlined above, pilots and engineers can optimize helicopter performance, ensure safety, and push the boundaries of rotary-wing aviation. **QuestionAnswer** What are the main principles behind helicopter lift generation? Helicopter lift is primarily generated by the rotor blades acting like rotating wings, producing lift through the aerodynamic force of airflow over their airfoil shape, following Bernoulli's principle and Newton's third law. How does blade pitch affect helicopter flight? Adjusting the blade pitch, or collective pitch, changes the angle of attack of the rotor blades, thereby controlling the amount of lift produced. Increasing pitch results in more lift, allowing ascent, while

decreasing pitch aids in descent. What is autorotation in helicopter aerodynamics? Autorotation is a state where the helicopter's rotor turns by aerodynamic forces rather than engine power, allowing safe descent and landing in case of engine failure by utilizing airflow to maintain rotor rotation. How does blade flapping influence helicopter stability? Blade flapping allows rotor blades to move up and down, balancing aerodynamic forces during flight. It helps equalize lift across the rotor disc, reducing vibrations and enhancing stability. What role does the tail rotor play in helicopter aerodynamics? The tail rotor counteracts the torque produced by the main rotor, preventing the fuselage from spinning. It also provides yaw control by varying the tail rotor thrust. How does blade twist improve helicopter performance? Blade twist gradually varies the blade's angle of attack from root to tip, optimizing lift distribution along the blade span, improving aerodynamic efficiency and reducing vibrations. What aerodynamic challenges are involved in helicopter rotor design? Design challenges include managing induced drag, blade vortex interaction, stall at high angles of attack, and vibration control, all of which require careful blade shaping and aerodynamic optimization. How does the advance ratio affect helicopter aerodynamics? The advance ratio, which is the ratio of forward speed to rotor tip speed, influences the aerodynamic forces on the rotor. Higher advance ratios can lead to asymmetric lift and stall on the advancing blade, affecting stability and control.

**Helicopter Aerodynamics: Unlocking the Secrets of Vertical Flight**

In the realm of aviation, helicopters stand out as marvels of engineering and aerodynamics, capable of vertical takeoff, hovering, and intricate maneuvers that fixed-wing aircraft cannot perform. At the core of these capabilities lie fundamental principles of helicopter aerodynamics—complex but fascinating phenomena that dictate how these machines generate lift, sustain stability, and maneuver through the air. Understanding these principles is essential not only for engineers and pilots but also for enthusiasts eager to grasp the science behind rotorcraft flight. In this comprehensive review, we explore the core concepts that underpin helicopter aerodynamics, dissecting each component to reveal how they work in harmony to achieve controlled, versatile flight.

--- **Fundamental Principles of Helicopter Aerodynamics**

Helicopter aerodynamics revolve around how rotor blades interact with the air to produce lift and thrust, enabling the craft to hover, ascend, descend, and move laterally or longitudinally. Unlike fixed-wing aircraft that rely on forward motion to generate lift, helicopters leverage their rotating blades—often called rotors—as spinning wings. The aerodynamic principles governing rotor operation are multifaceted, involving complex flow patterns, blade motion, and the interaction with the surrounding airflow.

**The Role of Rotor Blades: The Heart of Helicopter Aerodynamics**

Rotor blades serve as rotating wings, with their shape, angle, and motion meticulously designed to produce the desired aerodynamic effects. The blades are primarily airfoils—structures shaped to generate lift efficiently through the flow of air over their surfaces. Key aspects of rotor blades include:

- **Airfoil Shape:** Similar to airplane wings, rotor blades have an airfoil cross-section designed to produce lift with minimal drag.
- **Blade Twist:** The blades are often twisted along their length so that the angle of attack varies from root to tip, compensating for differences in relative airflow caused by rotation.
- **Blade Pitch Control (Collective and Cyclic):** Adjustments to the blade pitch allow pilots to control lift and maneuverability:
  - **Collective Pitch:** Changes the pitch angle of all blades simultaneously, controlling overall lift.
  - **Cyclic Pitch:** Varies the pitch angle cyclically as blades rotate, enabling directional control.

--- **Key Aerodynamic Phenomena in Rotor Operation**

Several

core aerodynamic phenomena come into play with helicopter rotors. Understanding these is vital to grasp how helicopters achieve stable flight and precise maneuvering.

**2.1 Lift Generation: The Blade Element Theory and Induced Flow** The fundamental task of a rotor blade is to generate lift, and this process is governed by classical aerodynamic theories such as blade element theory. This approach divides each blade into small sections, analyzing the forces on each segment to understand the overall lift production. Blade Element Theory simplifies the analysis by considering the following:

- The blade is segmented into small elements along its span.
- Each element acts like a small airfoil, generating lift based on the local angle of attack, airspeed, and airfoil shape.
- The total lift is the sum of the contributions from all elements.

**Induced Flow and Downwash** As blades generate lift, they impart a downward velocity component to the air—known as downwash or induced flow—which in turn influences the lift capacity. The interaction between the rotor and the airflow creates a feedback loop, where increased lift results in greater downwash, affecting the effective angle of attack and efficiency.

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**2.2 Principles Of Helicopter Aerodynamics 6 The Momentum Theory and Power Requirements** The momentum theory, or actuation theory, complements blade element theory by focusing on the energy transfer between the rotor and the airflow.

- **Thrust and Power:** To produce a certain thrust (lift), the rotor must impart momentum to the air, which requires power.
- **Induced Power:** The power needed to accelerate air downward, creating the lift force.
- **Profile Power:** The power lost overcoming blade drag and profile drag.

Understanding these power components helps in optimizing rotor design for efficiency and performance, balancing the trade-offs between lift, power consumption, and noise.

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**Advancing Concepts in Helicopter Aerodynamics** Beyond basic lift and power, several advanced aerodynamic effects and control mechanisms influence helicopter performance.

**2.1 Blade Tip Vortices and Tip Losses** Blade tip vortices are swirling air masses that form at the tips of rotor blades due to pressure differences between the upper and lower surfaces.

- These vortices cause tip losses, reducing the efficiency of lift generation.
- Design modifications, such as winglets or tip shapes, aim to mitigate vortex strength and improve aerodynamic efficiency.

**2.2 Hover vs. Forward Flight Aerodynamics** Helicopter aerodynamics differ substantially between hover and forward flight:

- **Hover:** The rotor must produce enough lift to counteract gravity, with airflow largely vertical and symmetrical.
- **Forward Flight:** The rotor disc becomes asymmetric, with the advancing blade experiencing higher relative wind speed, creating a phenomenon called dissymmetry of lift.
- To compensate, helicopters use blade flapping and cyclic pitch adjustments to balance lift across the rotor disc.
- The transition from hover to forward flight involves complex aerodynamic interactions, including the development of a retreating blade stall if not managed properly.

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**Control Principles: Managing Aerodynamics for Maneuverability** Helicopter pilots manipulate aerodynamic forces through control inputs, primarily via the cyclic, collective, and anti-torque pedals.

**3.1 The Cyclic Control: Controlling Direction** The cyclic adjusts blade pitch cyclically during each rotation, tilting the rotor disc to produce a net force in a desired direction.

- By increasing the pitch on one side of the rotor disc and decreasing it on the opposite, the helicopter tilts and moves laterally or longitudinally.
- Aerodynamically, this creates an asymmetric lift distribution, causing the craft to accelerate in that direction.

**3.2 The Collective Control: Vertical Lift Management** The collective pitch control changes the angle of attack for all blades simultaneously.

- Increasing collective pitch increases overall lift, enabling ascent.
- Decreasing it results in descent.
- The change in blade pitch affects the induced flow and overall

aerodynamics, requiring compensation to maintain stability. 3.3 Anti-Torque and yaw control Since the main rotor's rotation produces torque, the helicopter must counteract this: - Anti-torque Principles Of Helicopter Aerodynamics 7 pedals adjust the pitch of a tail rotor or other anti-torque system. - The aerodynamics of the tail rotor generate lateral thrust to counteract main rotor torque, allowing controlled yaw movement. --- Innovations and Aerodynamic Challenges While helicopter aerodynamics are well-understood, ongoing innovations aim to improve efficiency, reduce noise, and enhance safety. 4.1 Variable Geometry and Blade Design Modern rotor blades incorporate: - Composite materials for strength and weight reduction. - Blade twist and camber adjustments to optimize aerodynamic performance across flight regimes. - Active control systems for blade pitch and twist adjustments during flight. 4.2 Reducing Vortex and Induced Drag Design strategies focus on: - Blade tip modifications to reduce vortex strength. - Active flow control techniques to manipulate airflow around blades. 4.3 Challenges: Stall, Vortex Ring State, and Retreating Blade Stall - Blade Stall occurs when airflow separates from the blade surface, reducing lift. - Vortex Ring State is a dangerous condition where the helicopter descends into its own downwash, causing loss of lift. - Retreating Blade Stall happens at high forward speeds when the retreating blade's relative airflow drops below stall speed. Addressing these challenges involves precise aerodynamic analysis and sophisticated control systems. --- Conclusion: The Science That Powers Vertical Flight Helicopter aerodynamics is a complex tapestry of physics, engineering, and innovation. From the fundamental principles of lift generation and induced flow to advanced control mechanisms and cutting-edge blade design, each element plays a vital role in the mastery of vertical flight. The interplay of forces, flow patterns, and control inputs demonstrates the sophistication required to keep helicopters aloft and maneuverable. As technology advances, so too does our understanding of these aerodynamic principles, promising safer, more efficient, and quieter helicopters in the future. Whether for rescue missions, passenger transport, or military applications, the principles of helicopter aerodynamics continue to be the backbone of this remarkable mode of transportation—an elegant blend of physics and engineering that unlocks the skies. helicopter lift, rotor blades, blade angle, induced drag, autorotation, helicopter stability, rotor thrust, aerodynamic forces, tail rotor, vortex theory

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helicopters are highly capable and useful rotating wing aircraft with roles that encompass a variety of civilian and military applications their usefulness lies in their unique ability to take off and land vertically to hover stationary relative to the ground and to fly forward backward or sideways these unique flying qualities however come at a high cost including complex aerodynamic problems significant vibrations high levels of noise and relatively large power requirements compared to fixed wing aircraft this book written by an internationally recognized expert provides a thorough modern treatment of the aerodynamic principles of helicopters and other rotating wing vertical lift aircraft every chapter is extensively illustrated and concludes with a bibliography and homework problems advanced undergraduate and graduate students practising engineers and researchers will welcome this thorough and up to date text on rotating wing aerodynamics

written by an internationally recognized teacher and researcher this book provides a thorough modern treatment of the aerodynamic principles of helicopters and other rotating wing vertical lift aircraft such as tilt rotors and autogiros the text begins with a unique technical history of helicopter flight and then covers basic methods of rotor aerodynamic analysis and related issues associated with the performance of the helicopter and its aerodynamic design it goes on to cover more advanced topics in helicopter aerodynamics including airfoil flows unsteady aerodynamics dynamic stall and rotor wakes and rotor airframe aerodynamic interactions with final chapters on autogiros and advanced methods of helicopter aerodynamic analysis extensively illustrated throughout each chapter includes a set of homework problems advanced undergraduate and graduate students practising engineers and researchers will welcome this thoroughly revised and updated text on rotating wing aerodynamics

this is a collection of ray prouty s columns from rotor and wing magazine from 1979 to 1992

basic helicopter aerodynamics is widely appreciated as an easily accessible rounded introduction to the first principles of the aerodynamics of helicopter

flight simon newman has brought this third edition completely up to date with a full new set of illustrations and imagery an accompanying website wiley com go seddon contains all the calculation files used in the book problems solutions ppt slides and supporting matlab code simon newman addresses the unique considerations applicable to rotor uavs and mavs and coverage of blade dynamics is expanded to include both flapping lagging and ground resonance new material is included on blade tip design flow characteristics surrounding the rotor in forward flight tail rotors brown out blade sailing and shipborne operations concentrating on the well known sikorsky configuration of single main rotor with tail rotor early chapters deal with the aerodynamics of the rotor in hover vertical flight forward flight and climb analysis of these motions is developed to the stage of obtaining the principal results for thrust power and associated quantities later chapters turn to the characteristics of the overall helicopter its performance stability and control and the important field of aerodynamic research is discussed with some reference also to aerodynamic design practice this introductory level treatment to the aerodynamics of helicopter flight will appeal to aircraft design engineers and undergraduate and graduate students in aircraft design as well as practising engineers looking for an introduction to or refresher course on the subject

this is a collection of the ray prouty s columns in rotor and wing and american helicopter society s vertiflite magazine from 1992 to 2004

this book gives an account from first principles of the aerodynamics of helicopter flight concentrating on the well known sikorsky configuration of single main rotor with tail rotor early chapters deal with the aerodynamics of the rotor in hover vertical flight forward flight and climb analysis of these motions is developed to the stage of obtaining the principal results for thrust power and associated quantities but the lengthy mathematical treatment of some textbooks is avoided later chapters turn to the characteristics of the overall helicopter its performance stability and control and the important field of aerodynamic research is discussed with some reference also to aerodynamic design practice the second edition has been revised to illustrate more fully the various features of rotor aerodynamics and helicopter design the helicopter is unique in its linking of the aerodynamic and mechanical features and a full appreciation of these air vehicles can only be achieved by understanding these interactions many of the extra figures illustrate the diversity in the design and operation of a helicopter and these differences are highlighted in the text the book is aimed initially at the needs of undergraduates and postgraduates however because of its conciseness it is likely to prove useful also to workers at any stage as a background to short industrial courses or for anyone needing a refresher course in the basics of the subject

since the original publication of bramwell s helicopter dynamics in 1976 this book has become the definitive text on helicopter dynamics and a fundamental part of the study of the behaviour of helicopters this new edition builds on the strengths of the original and hence the approach of the first edition is retained the authors provide a comprehensive overview of helicopter aerodynamics stability control structural dynamics vibration aeroelastic

and aeromechanical stability as such bramwell s helicopter dynamics is essential for all those in aeronautical engineering the single volume comprehensive guide for anyone working with helicopters written by leading worldwide experts in the field

this book is developed to serve as a concise text for a course on helicopter aerodynamics at the introductory level it introduces to the rotary wing aerodynamics with applications to helicopters and application of the relevant principles to the aerodynamic design of a helicopter rotor and its blades the basic aim of this book is to make a complete text covering both the basic and applied aspects of theory of rotary wing flying machine for students engineers and applied physicists the philosophy followed in this book is that the subject of helicopter aerodynamics is covered combining the theoretical analysis physical features and the application aspects considerable number of solved examples and exercise problems with answers are coined for this book this book will cater to the requirement of numerical problems on helicopter flight performance which is required for the students of aeronautical aerospace engineering salient features to provide an introductory treatment of the aerodynamic theory of rotary wing aircraft to study the fundamentals of rotor aerodynamics for rotorcraft in hovering flight axial flight and forward flight modes to perform blade element analysis investigate rotating blade motion and quantify basic helicopter performance

this volume is an excellent introduction to the aerodynamics of helicopters basic helicopter aerodynamics provides an account of the first principles in the fluid mechanics and flight dynamics of single rotor helicopters the text is intended to provide in a short volume an introduction to the theory of rotary wing aircraft for use by undergraduate and graduate students while providing a detailed description of the physical phenomena involved the text assumes that the reader already has some knowledge of differences between the fixed and rotary wing aircraft many diagrams drawings graphs and representative sets of data augment the text

divclear concise text covers aerodynamic phenomena of the rotor and offers guidelines for helicopter performance evaluation originally prepared for nasa prefaces new indexes 10 black and white photos 537 figures div

this is a collection of the columns ray prouty wrote for the american helicopter society from 1992 2013 it covers a wide variety of helicopter related engineering subjects

basic helicopter aerodynamics is widely appreciated as an easily accessible rounded introduction to the first principles of the aerodynamics of helicopter flight simon newman has brought this third edition completely up to date with a full new set of illustrations and imagery an accompanying website wiley com go seddon contains all the calculation files used in the book problems solutions ppt slides and supporting matlab code simon newman addresses the



unique considerations applicable to rotor uavs and mavs and coverage of blade dynamics is expanded to include both flapping lagging and ground resonance new material is included on blade tip design flow characteristics surrounding the rotor in forward flight tail rotors brown out blade sailing and shipborne operations concentrating on the well known sikorsky configuration of single main rotor with tail rotor early chapters deal with the aerodynamics of the rotor in hover vertical flight forward flight and climb analysis of these motions is developed to the stage of obtaining the principal results for thrust power and associated quantities later chapters turn to the characteristics of the overall helicopter its performance stability and control and the important field of aerodynamic research is discussed with some reference also to aerodynamic design practice this introductory level treatment to the aerodynamics of helicopter flight will appeal to aircraft design engineers and undergraduate and graduate students in aircraft design as well as practising engineers looking for an introduction to or refresher course on the subject

although a number of texts on helicopter aerodynamics have been written few have explained how the various theories concerning rotorborne flight underpin practical flight test and evaluation this book combines theoretical information on aerodynamics stability control and performance with details of evaluation methodologies and practical guidance on the conduct of helicopter flight tests for each topic the relevant theory is explained briefly and followed by details of the practical aspects of testing a conventional helicopter these include safety considerations planning the tests the most efficient way to conduct individual flights where possible typical test results are presented and discussed the book draws on the authors extensive experience in flight test and flight test training and will appeal not only to professionals working in the area of rotorcraft test and evaluation but also to helicopter pilots rotorcraft designers and manufacturers and final year undergraduates of aeronautical engineering

the book contains the principles of helicopter flight special characteristics of the main rotor and its function in autorotation axial and oblique flow regimes of vertical and horizontal flight climb and descent takeoff and landing balance stability and control of the helicopter and their acting aerodynamic forces author

first published in 1952 by macmillan

aerodynamic research relating to modern helicopters includes the study of three dimensional unsteady nonlinear flow fields a selective review is made of some of the phenomenon that hamper the development of satisfactory engineering prediction techniques but which provides a rich source of research opportunities flow separations compressibility effects complex vortical wakes and aerodynamic interference between components several examples of work in progress are given including dynamic stall alleviation the development of computational methods for transonic flow rotor wake predictions and blade vortex interactions author

abstract the flow around a helicopter is very complex it becomes much more complex when it comes close to the ground the presence of the ground changes the aerodynamic characteristics of the rotor and the flow environment becomes much more complex compared with that of flight out of ground effect and hence the behavior of the rotor wake in the vicinity of the ground is challenging to predict under in ground effect conditions the wake collides with the ground and causes a significant perturbation to the flow near the blade significant interactions between the main rotor wake and the ground have been associated with the formation and passage of the ground vortex in forward flight the presence of a ground vortex affects the handling qualities of the helicopter the aim of this research is to capture the physics of the flow features and dynamics of ground effect flows around a rotorcraft provide an understanding of the rotor wake vortices near the ground and generate rigorous models to accurately predict handling qualities loads and moments acting on the rotor and the power requirements the wake structure after periodicity is reached is obtained for hover and different forward flight speeds also the nature of the flowfield as well as the formation of the ground vortex is understood by obtaining the velocity contours on a longitudinal plane containing the rotor blade after periodicity is obtained the unsteadiness in the velocities is quantified by obtaining the rms deviation in velocities on different planes containing the tail rotor around the rotor disk simulating the various kinds of flight thrust and power requirements on the rotor disk have been predicted and have been successfully validated by comparison with experimental results obtained from Georgia Institute of Technology

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